

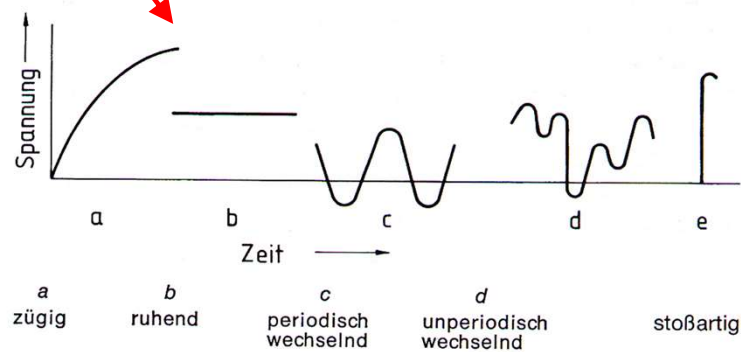
Why everything gets broken or material scientific consideration of damages

Thomas Gietzelt / IMVT-MAT



Causes of failure

- Design / calculation / manufacturing or material defects
- Material fatigue
- Fatigue in combination with additional reasons (e.g. corrosion)
- Insufficient maintenance / service
- Improper use / overload
- Human failure



Design-Issues: Collapse of the Tacoma-Narrows-Bridge 07.11.1940

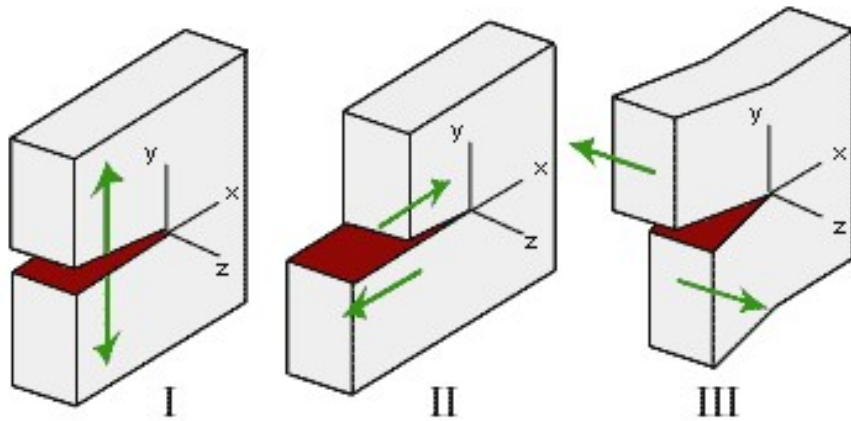
<https://www.youtube.com/watch?v=3mclp9QmCGs>
(starting at 01:00)



Reasons:

- Terrain profile (coast line formed nozzle)
- Rigidity of the bridge box too low, wrong self frequency calculation (resonance)
- It is impressive, what materials resist before failure...

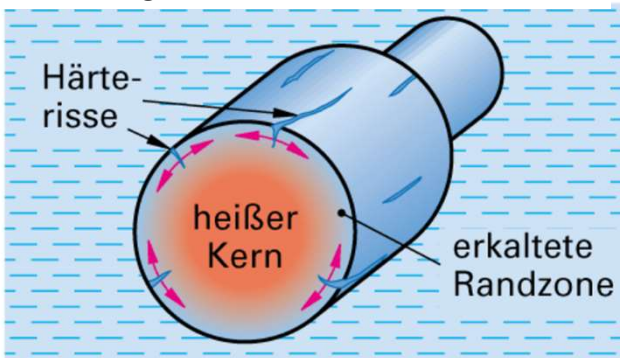
Crack opening modes I-III



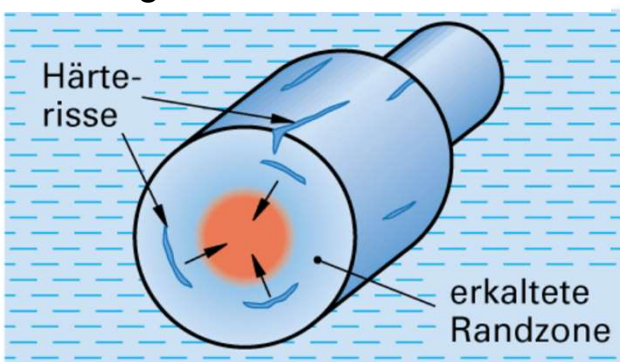
- Mode I: simple crack opening; symmetrical lift-off of the crack edges.
- Mode II: Longitudinal shear; sliding of crack surfaces in the crack plane.
- Mode III: Transverse shear; displacement of the crack surfaces transverse to the direction of the crack

Manufacturing/Processing: Crack initiation

1st phase: Hot core impedes shrinkage of shell



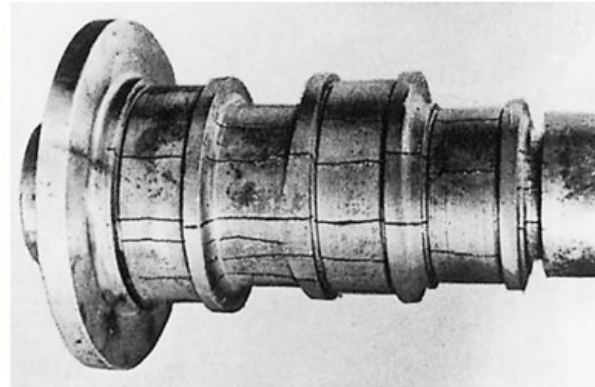
2nd phase: rigid shell impedes shrinkage of core



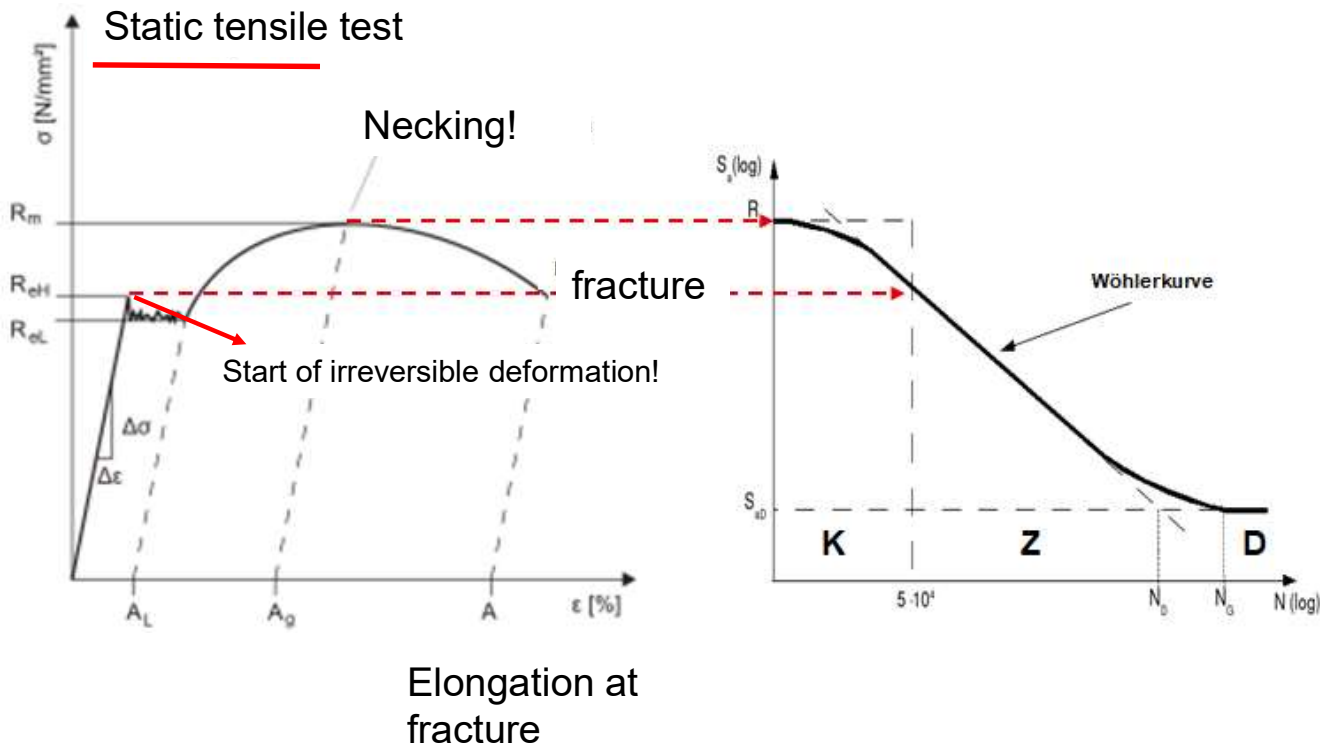
Hardening cracks due to volume change of austenite (fcc) and martensite (tetragonally distorted bcc), superimposition with thermal shrinkage

- Dye penetrant test
- Eddy current testing
- Depending on position: possibly not all cracks detected....

=> Superimposition of residual stress and external stresses (shot peening of surfaces)

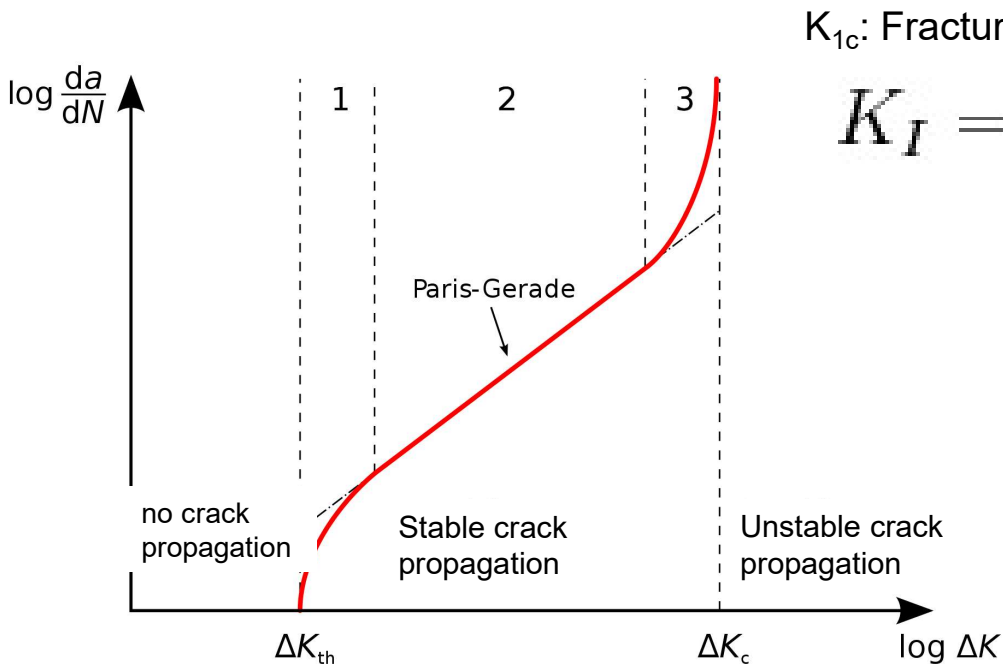


Fatigue resistance (Criterion: Fracture!)



- max. tolerable stress depends on the number of load cycles n , (depends on geometry!)
- **Short-term strength K** (100-30.000 load cycles $\approx R_e$)
- **Creep strength Z** (max. $2 \cdot 10^6$ load cycles \Rightarrow often not enough!)
- **Fatigue strength D** $\approx 10^6$ Lastspiele
- For $n \gg 10^6$ (10^8 - 10^9) few experiences, cannot be verified by tests (duration)
- **Engineering:** Mostly creep strength \Rightarrow Monitoring required!
- Crack growth & crack propagation models
- Sometimes: superimposition by unexpected incidents \Rightarrow failure

Why there is only a 5 year warranty against failure on aluminum bicycle frames or: there are no components without any cracks....



K_{1c} : Fracture toughness or critical stress intensity factor

$$K_I = \sigma \cdot \sqrt{\pi \cdot a} \cdot f$$

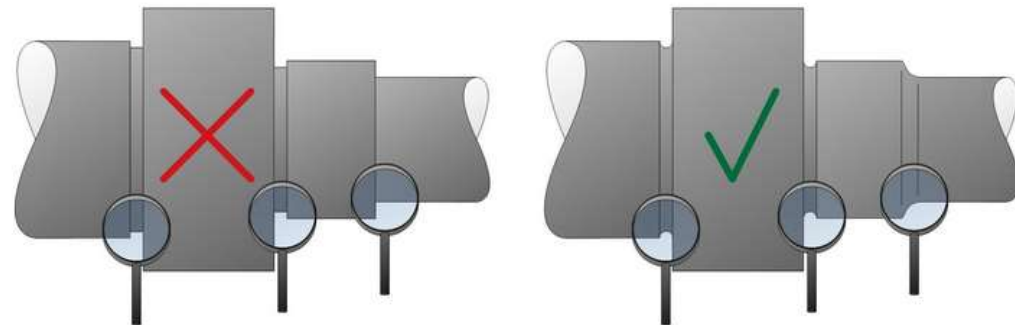
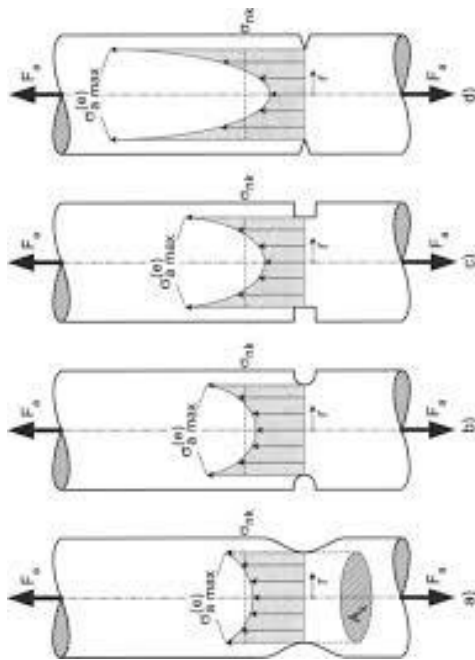
- σ Stress [MPa]
- a crack length [m]
- f geometric factor
- K_{1c} Fracture toughness [MPa*m^{1/2}]

K_{1c} depends on:

- Material
- **Design**
- **Surface roughness & surface state**
⇒ ceramic materials!

(Shot peening => Compressive residual stresses)

Impact of Design - Load optimized Design



Railway: Historical development of wheelset safety

Jahr	Wellenbrüche in Deutschland	Laufleistung zwischen zwei Wellenbrüchen in Deutschland
1880	80	100 Mio. km/Wellenbruch
1900	94	256 Mio. km/Wellenbruch
1930	88	414 Mio. km/Wellenbruch
1940	277	380 Mio. km/Wellenbruch
1955	225	117 Mio. km/Wellenbruch
1960	57	465 Mio. km/Wellenbruch
1970	35	826 Mio. km/Wellenbruch
1980	7	3.972 Mio. km/Wellenbruch
1990	7	3.866 Mio. km/Wellenbruch
2006	9	4.383 Mio. km/Wellenbruch
2007	4	10.099 Mio. km/Wellenbruch
2008	9	4.513 Mio. km/Wellenbruch
2009	1	39.040 Mio. km/Wellenbruch

Durchschnitt Deutschland 2006 – 2009

6.953 Mio. km/Wellenbruch

Durchschnitt EU-Bahnen 2006 – 2009

1.983 Mio. km/Wellenbruch

Quelle: Auswertung der statistischen Jahrbücher und ERA-Report durch BVV

=> Mileage between two damages at an all-time high!

ICE-Disaster Eschede 03.06.1998 - A sequence of unfortunate events



- *Composite wheel: Fracture of wheel rim due to **material fatigue**.*
- *Wheel rim made of rolled and tempered carbon steel, joined to the wheel disc via rubber sleeve.*
- *At the wheel contact point, the wheel rim is bent inward => tensile bending stresses in the circumferential direction at inner side.*
- *Per revolution: **change of stress tension** ⇔ **compression**.*
- ***Amplitude increased by wear and runout.***
- *Fatigue failure after 1.8×10^6 km (**$6,2 \times 10^8$ load cycles**).*
- *Broken wheel wedged into the wheel control arm. Fracture pushed wheels out of rails.*
- *Intact wheel hits a switch that had just been passed and changes it. The following wagons derailed. A bridge pier is broken down. The collapsing bridge buries one wagon. The remaining wagons are pushed by the heavy rear powerhead.*
- *101 dead , more than 100 people injured.*
- ***Runout was noticed at last maintenance but considered still uncritical => crack growth faster than expected.***
- ***Consequence:** Reduction of maintenance intervals of wheelsets from 300,000 => 60,000 => 30,000 km (capacity problem for maintenance)*

Different Railroad Wheels

Composite wheels consist of a wheel body and a wheel rim (different steel alloys).
Wheel rims are shrunk => tensile stresses.

Full wheel : forged from one piece

- By heat treatment, surface is hardened (if worn out, the entire wheel must be replaced).
- Full wheel is pressed onto the wheel shaft. =>
- Design of tolerances based on **operating experience**.

Quality: Random samples destructively tested; **100% NDT**.
=> Highest stresses occur in tight curves and switches at low speed; never at high speed when driving straight !!!

=> Most of damages with "fatigue fractures" is due to corrosion (**mixed loading; failure hardly predictable**).



Wheelset shaft fracture at ICE 3 on 09.07.2008 in Köln



Broken axle shortly before Cologne main station; train derails on departure (!)

Reason: Manufacturing-related material inclusions in the wheelset shaft.

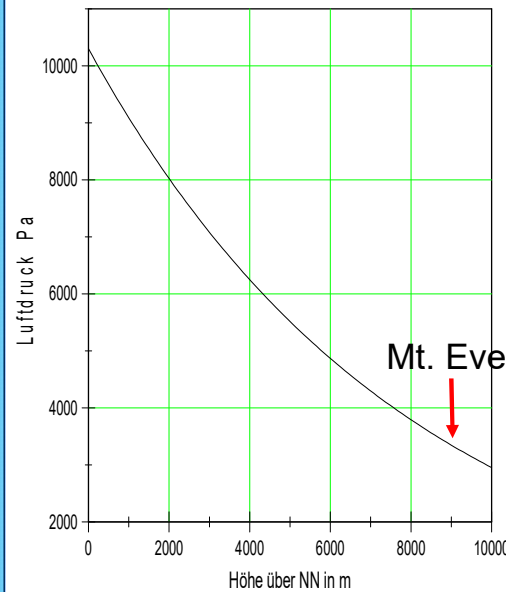
Only fortunate circumstances prevent an accident; cruising speed shortly before 330km/h!

Consequence: Reduction of maintenance intervals of wheelsets from 300,000 => 60,000 => 30,000 km (capacity problem for maintenance)

Conflict between Technology ↔ Society

- German railroads: Operating experience for mileages from 2.5 million km to 10 million km without crack detection.
 - Load cycles of approx. $9 \cdot 10^8$ or $4 \cdot 10^9$ (similar to crankshafts and generator shafts).
 - No noticeable accumulation of damage at high mileages.
 - Predominant number of damages with "fatigue fractures" due to corrosion. Gesellschaft erwartet technische Höchstleistungen ohne jegliches Risiko („Vollkasko-Gesellschaft“)
1. There is no such thing as operation without any failure!
 2. There is always a certain probability of occurrence for failure
 3. The maintenance is designed according to it. **Question:** Are the assumptions o.k.?
 4. Consideration & social acceptance of the residual risk

Barometric Elevation Formula



Barometrische Höhenformel:

$$p(h) = p_0 \cdot e^{-\frac{\rho_0 \cdot g \cdot h}{p_0}}$$

Mt. Everest: 8848m

Aloha-Airlines-Flight 243, (Boeing 737-200, 28.4.88)



Reason:

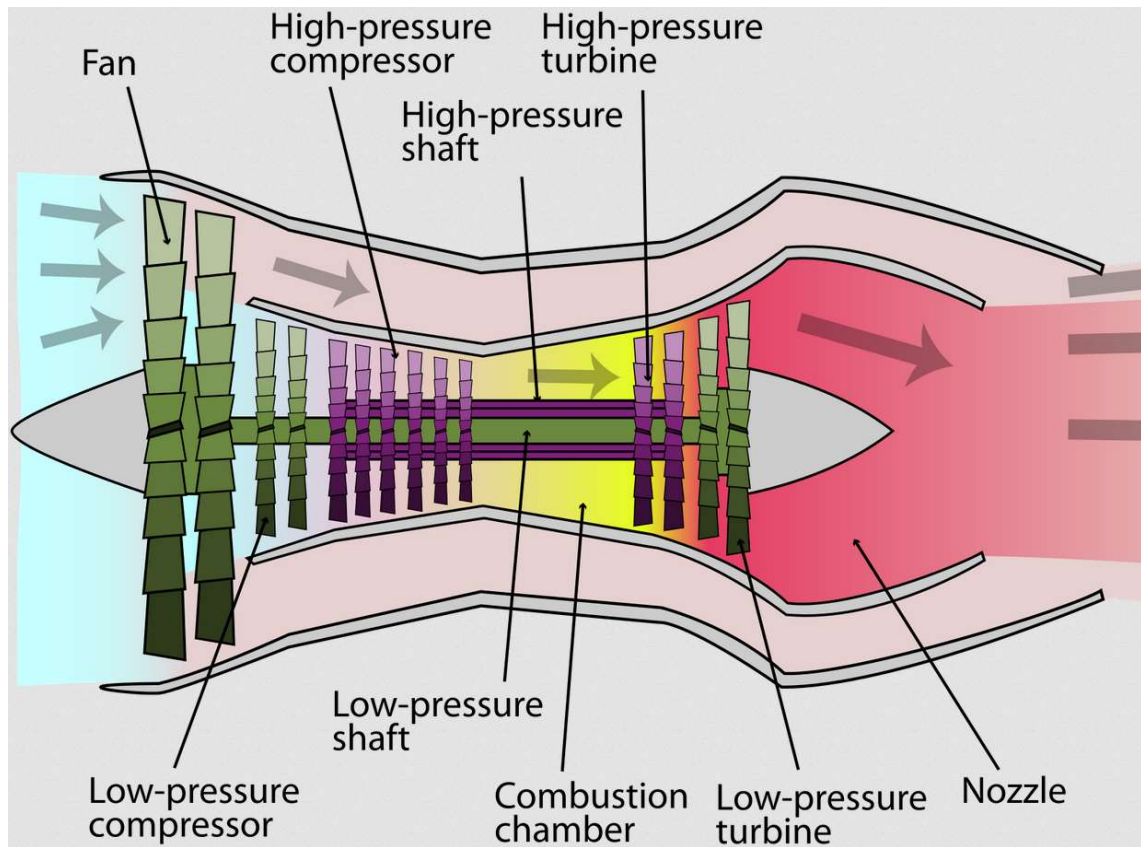
- Island-hopping (short range)
- after 90.000 cycles (designed for 50.000 only *)
- corrosion (salt spraying) +
- fatigue (Δp)
- Crack propagation between rivet holes in the hull @ 7300 m

Consequences:

- 1 flight attendant sucked off (†)
- 65 people injured, 7 of it severe

*) D. Rataj, „Ermüdungsfestigkeit“, Springer, ISBN 978-3-662-07108-3, S.3

Turbofan



- Very efficient & quiet (housing)
- Almost all commercial aircraft
- By-pass ratio up to 1:12
- GE9X: 597 kN thrust, Fan- \varnothing =3,4 m

Flight QF32 (A 380, 4.11.10) and United-Flight UA328 (Boeing 777, 20.2.21 and some more....)

A few years ago, engine explosion was a case-1 accident, resulting in a total loss of the aircraft

Today, the most frequent reason of crashes is operator failure

QF 32:

- Fatigue fracture of an oil rail in the RR Trent-900 engine
- Turbine disk detached from drive shaft due to oil fire.
- Engine casing, left wing, 2 tanks & hydraulic lines penetrated.

UA328

- two fan blades broken



Composite Materials: Crash of AA 587 (Airbus A300, 4.11.01)



- Generally: Problems with NDT, no ultrasonic testing possible.
- Airbus A300 flew through wake vortices of a previously launched Boeing 747 after takeoff in N. Y.
- Copilot reacted with full deflections of the rudder in both directions (hydraulically operated, no haptic feedback!).
- Load limits were exceeded, rudder broken.



Coal-fired Power Station

Steam temperature: up to 600°C,
pressure: 80-300 bar (8-30 MPa),
Steam throughput: up to 350 t/h
Efficiency: up to 47% (Nordjylland 3,
Danmark, 411 MW)

Boiler house of power station Staudinger
(Großkotzenburg, 12.05.2004) after „steam leakage“



Conclusions

- Often technical equipment is operated in the range of creep rupture strength
- Engineering monitoring and maintenance plans are necessary and
- Underlying assumptions of load cases & failure mechanisms must be appropriate
- Otherwise, catastrophic failure is possible
- Usually engineering designs behave well and endure more than what they were designed for
- There is no technology without residual risks....
- Nowadays, safety is at a level never reached before
- However, due to global reporting, accidents are becoming more prominent in the public's mind.